

**RECENT RADIATION EFFECTS ACTIVITIES AT JPL:
Coping with COTS**

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ABSTRACT

The radiation effects and testing programs in place at the Jet Propulsion Laboratory support the insertion of traditional, advanced and emerging microelectronic and photonic technologies in NASA systems through research and testing that enhances the radiation hardness assurance (RHA) of these technologies. With the decreasing availability of radiation hardened electronics and the new NASA paradigm of faster, more aggressive and less expensive space missions, there has been an increasing emphasis on using high performance commercial microelectronic parts and circuits in NASA spacecraft.

The use of commercial parts and circuits (often referred to as commercial off-the-shelf (COTS)) in space systems poses many potential problems, especially with regard to RHA. The rapid evolution of traditional types of circuits, such as microprocessors and memories, is driven by intense competition in the private sector, and these advances, primarily due to scaling effects, often result in increased vulnerability to radiation. In addition, radiation testing which reveals that a particular generation of parts is appropriate for certain dose ranges may be rendered irrelevant by the introduction of next-generation versions and the disappearance of the radiation tolerant parts from the market place. The introduction of new and emerging technologies that promise greater performance without increased power, weight or volume may have completely unknown radiation effects behavior that must be established through testing. These issues and others will be addressed and discussed within the context of recent radiation effects research and testing activities at JPL, that include the following:

1. Within the last few years it has become apparent that certain linear bipolar integrated circuits, such as q-amps, comparators and voltage references, are more sensitive to total ionizing dose (TID) at the low dose rates characteristic of the space environment than they are at moderate to high dose rates traditionally employed for total dose RHA testing. This RHA problem is exacerbated by the difficulties associated with doing realistic testing at low dose rates for fast, aggressive space projects. Data demonstrating the enhanced low dose rate (ELDR) effect will be presented along with results of efforts to develop a satisfactory RHA test based on moderate to high dose rate exposure at elevated temperatures.

2. Recent failures of optocouplers on the Topex Spacecraft have highlighted the importance of addressing RHA issues for non-traditional electronics such as optoelectronic devices. LEDs, photodetectors and optocouplers can vary widely in their response to radiation.

Furthermore, displacement damage effects due to protons can be important, as demonstrated by Topex, even though proton fluences are not particularly high for LEO and GEO spacecraft, relative to the magnitude usually required to cause problems in III-V devices. These problems are even more severe in optocouplers that employ phototransistors because of their potential sensitivity to both ionizing radiation and displacement damage. Recent results of proton damage studies of these devices, including comparisons with Topex data, will be presented.

3. High performance scientific instruments aboard spacecraft often require the use of high resolution, high speed analog to digital converters (ADCs). Unfortunately, because there are essentially no radiation tolerant devices of this type available, instrument and spacecraft designers often have to select devices that will not tolerate the anticipated environment. RHA testing of these devices is difficult for both TID and single event effects (SEE), especially when they are BiCMOS. Because of the ELDR effect a BiCMOS ADC can fail by different mechanisms depending on the conditions of radiation exposure. Results demonstrating these effects will be discussed in our presentation.

4. Field programmable gate arrays are particularly attractive for spacecraft use for several reasons. They often allow a significant reduction in the number of parts used on a board, and they can be individually personalized so that a large common buy of parts with established reliability and RHA can be used for several different applications. However, these devices exhibit a variety of radiation effects including energetic ion-induced linking of anti-fuse links. Recent SEE and TID results will be presented for Actel FPGAs.

5. Highly autonomous, heavily instrumented spacecraft require large amounts of mass memory to support on-board data processing and storage. Recently, on the Cassini spacecraft for example, this requirement has manifested itself in the use of large numbers of DRAMs for multi-gigabit mass storage as a replacement for tape recorders. TID and SEE studies of DRAMs have revealed interesting effects including the use of retention time as a highly sensitive measure of TID response and the observation of what appear to be two types of hard errors induced by energetic ions. These results will be discussed in our presentation.

6. An exciting new class of devices are microelectromechanical systems (MEMS) in which microelectronics are combined with miniaturized mechanical devices on a single chip. These devices allow reductions in size of a wide variety of functional elements that are used on spacecraft. MEMS devices can be sensitive to radiation for two reasons: 1) the electronics on the chip can exhibit radiation responses, and 2) certain types of mechanical devices, such as cantilevers that depend on capacitive charge for spatial displacement, can also conceivably be sensitive to radiation. We will present recent results of total dose studies of microaccelerometers that suggest that both types of radiation sensitivities are exhibited by these devices.

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*This information is also available on the ESTEC Web Server
(<http://www.estec.esa.nl>)*

3rd ESA ELECTRONIC COMPONENTS CONFERENCE
21 -25 April 1997
ESTEC,NOORDWIJK, THE NETHERLANDS

Introduction

The third ESA Electronic Components Conference, will be held in a time of fast and turbulent change for space programmes in the way that they procure and assure components. In a diminishing market for high reliability components new methods are sought and developed and existing ways of work and management are questioned and changed. The conference will reflect these developments in assurance, management and standardisation methods within European and international space programmes. Advancements in component technologies and physics, testing and measurement techniques and tolerance to the space environment will also form part of the conference programme.

Within the scope of the conference all major activities relating to selection, procurement, control and utilisation of electronic, electrical and electromechanical components for space application will be addressed with particular emphasis on assurance and standardisation methodologies, component cost matters, new component technologies and assessment methods.

Invited papers will highlight the major topics and Panel/Round Table discussions will be held on key issues. Tutorial and poster sessions are foreseen.

In the same way as in the previous successful conferences, this third ESA Electronic Components Conference will provide ample opportunities for project and corporate managers, engineers and scientists to exchange ideas and information on electronic component subjects.

The Conference will be organised in themes covering the following subjects:

THEME A: COMPONENT ENGINEERING AND TECHNOLOGY

Theme A will discuss component technology and physics, test and inspection methodology, applications, failure modes etc. The following will be addressed:

- new component technologies and applications,
- new evaluation methods and efficiency of testing techniques,
- reliability studies and evaluation of failure modes,
- component evaluation and approval programmes,
- component failure experience in space systems.

THEME B: COMPONENT PACKAGING AND ASSEMBLY

Theme B will discuss the rapid development in component packaging techniques and related assembly methods. Special attention will be given to the following:

- high pin count packages for complex integrated circuits,
- usage of plastic packages for space applications,
- packages for microwave and MMIC components,
- surface mount devices and technologies,
- packages for high power dissipation devices,
- multichip modules,
- approaches for chips, wafers and known good die.

THEME C: COMPONENT STANDARDISATION, SPECIFICATION AND ASSESSMENT

Theme C will discuss the developments in international and national component standardisation and specification systems and address new component assessment methods. The following will be addressed:

- international and national standards and specification systems,
- best commercial practices, synergy with other markets,
- TQM, QML, technology and capability approval,
- certification and assurance concepts,
- commercial part adaptation to space utilisation.

THEME D: COMPONENT PROCUREMENT, COST AND PROGRAMMATIC ASPECTS

Theme D will address problems in availability and cost of (key) components for space programmes ranging from microsatellites to space stations as well as organisation and control of component procurements and storage.

The following items are of interest:

- type reduction, PPL and part approval procedures, cost of ownership, component costs and procurement costs,
- component stores and test and assembly houses, non conformance and alert systems,
- procurement approach vs system and mission complexity,
- component data management and dissemination

THEME E: SPACE RADIATION EFFECTS

Theme E will discuss the influence of space radiation on electronic components and will address:

- impact of radiation on component reliability,
- radiation countermeasures, radiation hardness assurance of components, packages for improvement of radiation tolerance cosmic ray, total dose and proton test data.

CALL FOR PAPERS

Authors are requested to prepare extended abstracts (2 pages) providing sufficient details to permit a meaningful review. In addition a summary of not more than 35 words requested for inclusion in the second announcement a preliminary programme.

Presentations will be selected by the Program Committee on the basis of the following:

- general interest of the subject,
- quality of the content,
- originality and innovativeness of the ideas.

Abstracts may be submitted:

- by using the form on the www, "to transfer or e-mail attachment (please use only plain text);
- by mail,
- by fax (followed by an original copy in the mail)

Abstracts should contain the following information:

- title of the paper (centred) and proposed Theme,
- author(s) name(s)
- affiliation of the author(s)

The full address of the contact person, including telephone/fax numbers and e-mail address should be given separately.